

**(a) Effective Date**

This airworthiness directive (AD) is effective October 13, 2026.

**(b) Affected ADs**

None.

**(c) Applicability**

This AD applies to Costruzioni Aeronautiche Tecnam S.P.A. (Tecnam) Model P2006T airplanes, certificated in any category, serial numbers (S/N) 001 through 434 and S/N 442.

**(d) Subject**

Joint Aircraft System Component (JASC) Code 2710, Aileron Control System.

**(e) Unsafe Condition**

This AD was prompted by a report of partial detachment of the left aileron due to a missing hinge pin. The FAA is issuing this AD to detect and address cracks, damage, or improper installation of the aileron hinges and cotter pins. The unsafe condition, if not addressed, could lead to detachment of an aileron, possibly resulting in loss of control of the airplane.

**(f) Compliance**

Comply with this AD within the compliance times specified, unless already done.

**(g) Required Actions**

(1) Within 25 flight hours or 30 days, whichever occurs first after the effective date of this AD, perform a one-time detailed visual inspection of the inboard and outboard aileron hinges of the left and right ailerons for damage (cracks, warping of the hinges or hinge pin, scratched paint, missing or broken torque stripe, etc.), and inspect the cotter pins installed at both ends of each hinge pin for improper installation and evidence of looseness, in accordance with steps N 4 and 5 of the Accomplishment Instructions in Tecnam Service Bulletin No. SB 995-CS-Ed. 1 Rev. 2, dated September 18, 2026 (Tecnam SB 995-CS-Ed. 1 Rev. 2).

(2) If, during the inspection required by paragraph (g)(1) of this AD, no discrepancy affecting a cotter pin or crack or damage of an aileron hinge is found, no further action is required by this AD.

(3) If, during the inspection required by paragraph (g)(1) of this AD, any discrepancy affecting a cotter pin is found, before further flight, replace and secure the cotter pin, in accordance with step N 6 of the Accomplishment Instructions in Tecnam SB 995-CS-Ed. 1 Rev. 2.

(4) If, during the inspection required by paragraph (g)(1) of this AD, any crack or damage is found on an aileron hinge, before further flight, contact the Manager, International Validation Branch, FAA; or

European Union Aviation Safety Agency (EASA); or Tecnam's EASA Design Organization Approval (DOA) for approved corrective instructions. If approved by the DOA, the approval must include the DOA-authorized signature. Accomplish the approved corrective instructions accordingly.

#### **(h) No Reporting Requirement**

Although Tecnam SB 995-CS-Ed. 1 Rev. 2 specifies to submit information to the manufacturer, this AD does not require that action.

#### **(i) Credit for Previous Actions**

This paragraph provides credit for actions required by paragraph (g) of this AD, if those actions were performed before the effective date of this AD using Tecnam Service Bulletin No. SB 995-CS-Ed. 1 Rev. 0, dated April 21, 2026, or Tecnam Service Bulletin No. SB 995-CS-Ed. 1 Rev. 1, dated August 25, 2026.

#### **(j) Alternative Methods of Compliance (AMOCs)**

The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (k)(1) of this AD and email to [AMOC@faa.gov](mailto:AMOC@faa.gov). Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

#### **(k) Additional Information**

(1) For more information about this AD, contact Anthony Kenward, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (817) 222-5152; email: [anthony.b.kenward@faa.gov](mailto:anthony.b.kenward@faa.gov).

(2) Material identified in this AD that is not incorporated by reference is available at the address specified in paragraph (l)(3) of this AD.

#### **(l) Material Incorporated by Reference**

(1) The Director of the Federal Register approved the incorporation by reference of the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Tecnam Service Bulletin No. SB 995-CS-Ed. 1 Rev. 2, dated September 18, 2026.

(ii) [Reserved]

(3) For Tecnam material identified in this AD, contact Costruzioni Aeronautiche Tecnam S.P.A., Via Maiorise, 81043 Capua CE, Italy; phone: +39 0823 997538; email: [technical.support@tecnam.com](mailto:technical.support@tecnam.com); website:

(4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 1100 Main, Kansas City, MO 64105. For information on the availability of this material at the FAA, call (817) 222-5110.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit [www.archives.gov/federal-register/cfr/ibr-locations](http://www.archives.gov/federal-register/cfr/ibr-locations) or email [fr.inspection@nara.gov](mailto:fr.inspection@nara.gov) .